

**Update to TSO-C161a Regarding Airborne Constraint Region**  
**7/2/2012**

Since the publication of TSO-C161a, RTCA has made additional changes to RTCA-DO-253C, further restricting the allowable airborne constraint regions of both double delta and early-minus-late correlators. RTCA has adopted these changes once it was determined the previously approved GBAS ground station would not be able to protect the complete airborne constraint region with the required level of integrity.

These changes in the allowable airborne user constraint region will be formally incorporated into the next version of RTCA Document DO-253C at a future date. In the meantime, it is recommended that TSO manufacturers include applicable part of these changes to their design as a deviation to their proposed TSO-C161a equipment, utilizing this update to document the basis for equivalent level of safety in accordance with 14 CFR 21.609(a). Should you have any questions, please contact Hamza Abduselam, AIR-130, at (202) 385-4688.

Proposed Changes

In order to reflect this reduction in the allowable airborne user constraint region, the following changes are necessary when using TSO-C161a:

- 1) Replace section 2.3.6.4.1 of RTCA DO-253C with the following

**2.3.6.4.1 GPS Tracking Constraints**

GPS satellites shall [LAAS-303] be tracked using either an early-minus-late or double delta delay lock loop discriminator.

For early-minus-late (E-L) delay lock loop (DLL) discriminator tracking of GPS satellites, the pre-correlation bandwidth of the installation, the correlator spacing (d), and the differential group delay shall [LAAS-091] be within the ranges as defined in Table 2-6 for the applicable GAEC and illustrated in Figure 2-3.

**Table 2-6 GPS Tracking Constraints for E-L DLL Discriminators**

Region (see Figure 2-3)	3 dB Pre- correlation bandwidth, BW	Average Correlator Spacing (d) [C/A chips]	Instantaneous Correlator Spacing (d) [C/A chips]	Differential Group Delay	Appli- cable GAEC
1	4<BW≤7 MHz	0.045-0.21	0.04-0.235	≤ 600 ns – D <sub>A</sub> – D <sub>C</sub>	C
2	7<BW≤16 MHz	0.045-0.21	0.04-0.235	≤ 150 ns – D <sub>A</sub> – D <sub>C</sub>	C & D
3	16<BW≤20 MHz	0.045-0.12	0.04-0.15	≤ 150 ns – D <sub>A</sub> – D <sub>C</sub>	C & D
4	20<BW≤24 MHz	0.08-0.12	0.07-0.13	≤ 150 ns – D <sub>A</sub> – D <sub>C</sub>	C & D

*Note:* D<sub>A</sub> is the differential group delay contribution of the antenna through the output of the pre-amp. D<sub>C</sub> is the differential group delay contribution of the installation specific connection between the antenna and the PAN equipment.

*Note:* Region 4 is not practical for airborne equipment that also track SBAS ranging signals when implemented using a common receiver front end for receiving the GPS and SBAS signals. This is because the SBAS

*tracking constraints given in Table 2-9 do not include bandwidths in Region 4 of Table 2-6.*

The instantaneous correlator spacing is defined as the spacing between a particular set of early and late samples of the correlation function. The average correlator spacing is defined as a one-second average of the instantaneous correlator spacing. The average applies over any one-second time frame.

The discriminator ( $\Delta$ ) shall [LAAS-092] be based upon an average of early-minus late samples with spacings inside the specified range. Either a coherent or a non-coherent discriminator may be used.

For Double Delta (DD) DLL discriminators of the type  $\Delta=2\Delta_{d1} - \Delta_{2d1}$  tracking GPS satellites, the pre-correlation bandwidth of the installation, correlator spacings ( $d_1$  and  $2d_1$ ) and the differential group delay shall [LAAS-093] be within the specified ranges as defined in Table 2-7 for the applicable GAEC and illustrated in Figure 2-3. Either a coherent or a non-coherent discriminator may be used.

*Note: These tracking constraints for DD DLL discriminators are different than the tracking constraints contained in DO-229() and DO-316().*

**Table 2-7 GPS Tracking Constraints for DD DLL Discriminators**

Region (see Figure 2-3)	3 dB Pre-correlation bandwidth, BW	Average Correlator Spacing ( $d_1$ and $2d_1$ ) [C/A chips]	Instantaneous Correlator Spacing ( $d_1$ and $2d_1$ ) [C/A chips]	Differential Group Delay	Appli- cable GAEC
1	$(-50*x)+12 < BW \leq 7$ MHz	0.1-0.16	0.09-0.18	$\leq 600$ ns $-D_A - D_C$	C
	$4 < BW \leq 7$ MHz	0.16-0.6	0.14-0.65		
2	$(-50*x)+12 < BW \leq$ $(133.33*x)+2.667$ MHz	0.07-0.085	0.063-0.094	$\leq 150$ ns $-D_A - D_C$	C & D
	$(-50*x)+12 < BW \leq 14$ MHz	0.085-0.1	0.077-0.11		
	$7 < BW \leq 14$ MHz	0.1-0.24	0.09-0.26		
3	$14 < BW \leq 16$ MHz	0.1-0.24	0.09-0.26	$\leq 150$ ns $-D_A - D_C$	C & D
	$14 < BW \leq (133.33*x)$ $+2.667$ MHz	0.085-0.1	0.077-0.11		

*Notes: (1) In the pre-correlation bandwidth column, x denotes the average correlator spacing.*

*(2)  $D_A$  is the differential group delay contribution of the antenna through the output of the pre-amp.  $D_C$  is the differential group delay contribution of the installation specific connection between the antenna and the PAN equipment.*

The differential group delay, which applies to the entire aircraft installed system, including that of the antenna ( $D_A$ ), any installation specific cabling or active devices ( $D_C$ ), and the RF front end of the PAN, must be bounded. However,

there is some flexibility in the apportionment of differential group delay among these components.

If the equipment uses an RTCA/DO-301 minimum standard compliant antenna,  $D_A$  is 25 ns. If the equipment uses a specific RTCA/DO-301 compliant antenna, it may take advantage of any reduced differential group delay (i.e.,  $D_A$  may be  $< 25$  ns).

In addition, an aircraft installation consisting of only cable and connectors may be assumed to have a differential group delay contribution ( $D_C$ ) of 0 ns. However, installations incorporating devices such as splitters or amplifiers may introduce additional differential group delay. The manufacturer may support such installations, but has no obligation to do so. In any event, the manufacturer must specify the maximum acceptable installation related differential group delay ( $D_C$ ). This limit should be defined in the installation instructions.

*Note: Equipment built to earlier versions of the document are assumed to support only  $D_C = 0$ . The original receiver requirements derivation included no installation allocation.*

The differential group delay is defined as:

$$\frac{1}{360} \cdot \left| \frac{d[\Phi(f_1)]}{df} - \frac{d[\Phi(f_2)]}{df} \right|$$

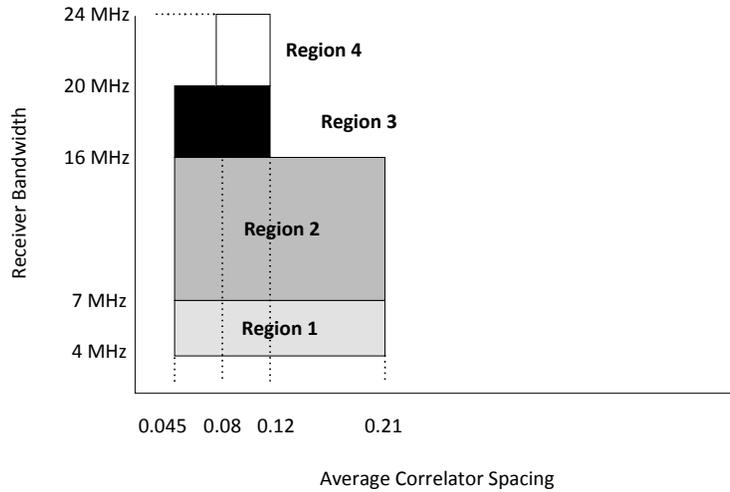
where:

$f_1$  and  $f_2$  are any frequencies within the 3 dB bandwidth of the pre-correlation filter.

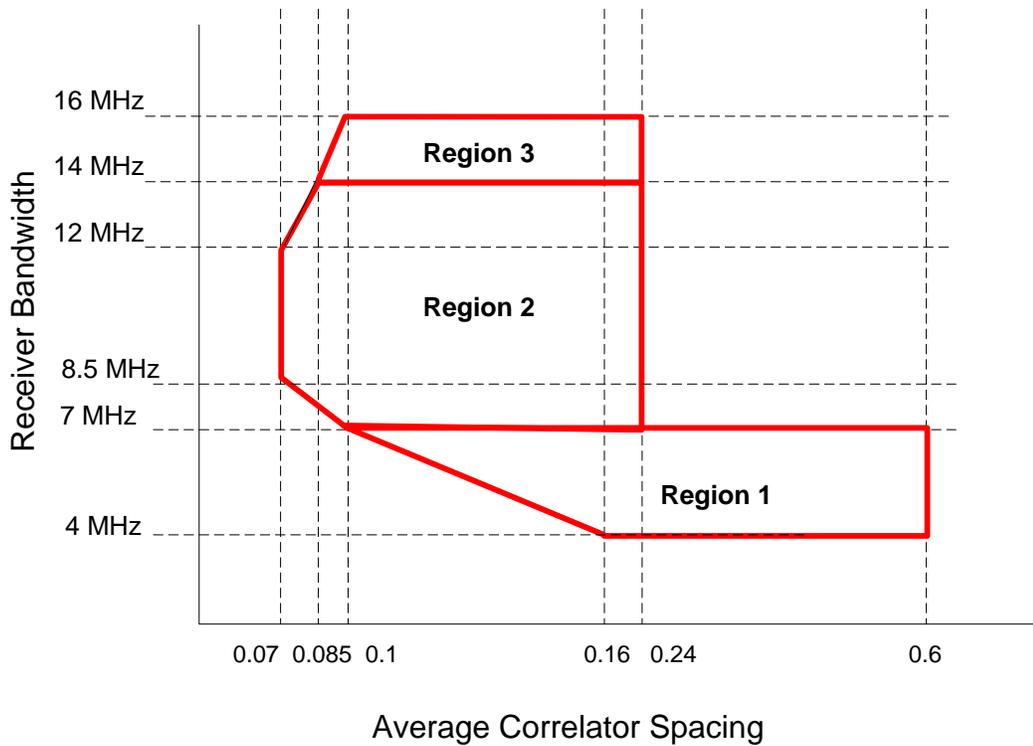
$\Phi(f)$  is the combined phase response of the equipment in degrees (excluding the antenna).

$f$  is the frequency in Hz.

For the DD DLL discriminators, the pre-correlation filter shall [LAAS-304] roll-off by at least 30 dB per octave in the transition band which starts at the -3dB points, and the resulting attenuation in the stop band shall [LAAS-396] be greater than or equal to 50dB (relative to the peak gain in the pass band) at frequencies more than 24 MHz from the band center.



**E-L Discriminator Tracking of GPS Satellites**



**DD Discriminator Tracking of GPS Satellites**

**Figure 2-3 Receiver Bandwidth vs. Average Correlator Spacing**

*Note: The technical implementation of the airborne receiver must be constrained to enable the LAAS ground system to effectively protect the airborne receiver from possible degradations in the GPS satellite signal. These constraints are described in terms of correlator spacing, receiver*

*bandwidth and receiver differential group delay. The satellite signal degradations considered in developing these constraints included:*

- a) Distorted satellite signal causing multiple correlation peaks*
- b) Correlator peak distortion due to code coherent spurious signals (such as reflected signals or code transition induced wave forms in the satellite)*
- c) Code coherent spurious signals distorted by RF filter differences*
- d) Flat correlation peaks causing excessive noise or drift*
- e) Discriminator behavior based on transient distorted satellite signal conditions.*

2) Delete Table 2-7 and its accompanying notes under the table in appendix 1 of TSO-C161a